## Space Solar Power (SSP) Systems Studies and Analysis

The Aerospace Corporation Sept 10-12, 2002

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#### Aerospace SERT Tasks

- Define innovative concepts, applications, and orbits
  - Work with NASA to define set of SSP concepts to be modeled
- Develop subsystem-level integrated SPS models
  - Enhancement of subsystem models
  - Development of new models, based on chosen concepts and applications
  - Development of up to 5 SPS models, each model for one concept and one application
  - Extended Effort: Development of laser SSP concepts and models
- Conduct subsystem-level trades to "optimize" each SSP system
  - In-depth trade space exploration
  - Determine key system and technology characteristics
- Conduct system-level trades between SSP concepts
  - Compare various SSP concepts for similar applications
- Generation of inputs to investment roadmaps
  - Integrate the requirements of the most promising concepts into a set of technology goals

#### Aerospace SSP Team

Study Lead Jay Penn

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Astrodynamics George Chao

Cost Vince Canales

Economics Jay Penn, John Skratt

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Systems Glenn Law, Christopher Taylor

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TT&C / C&DH / WPT John O'Donnell, Ed Olson

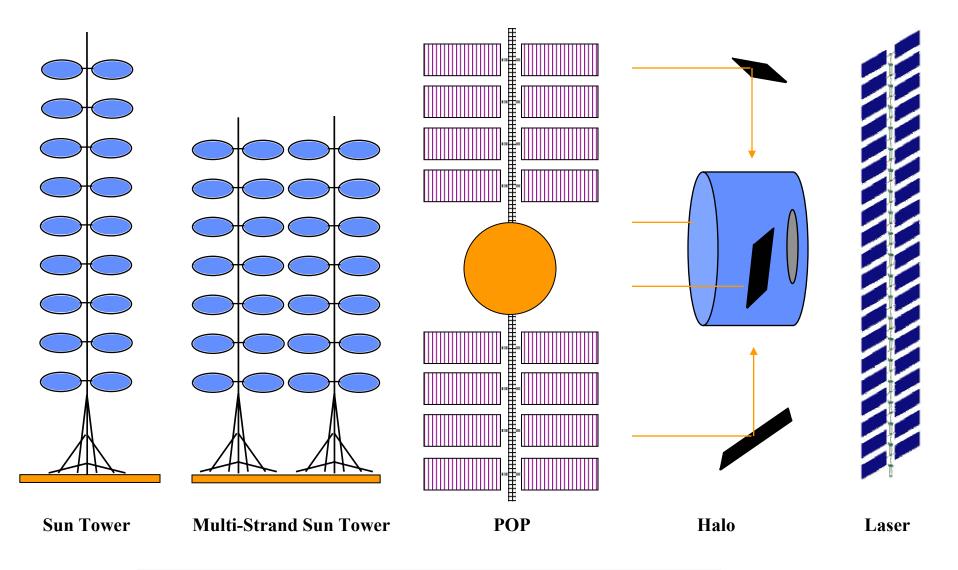


# Summary of Aerospace SERT Accomplishments

- Performed detailed modeling on selected SSP concepts
  - Sun Tower concept
  - Multi-Strand Sun Tower concept
  - Perpendicular to Orbit Plane (POP) concept
  - Halo concept
  - Laser concept
  - Microwave demonstrator (MSC-1) concept
  - Laser demonstrator (MSC-1) concept
- Incorporated SSP subsystem model enhancements
  - Enhanced ADACS, Economics, Power, Structures, Thermal, WPT models
  - Developed Cost, Ground System, Manufacturing, Orbit Analysis, PMAD,
     Propulsion models
- Refined Halo and Laser design, tradespace, related concepts, and applications
- Contribute to WPT, SIWG, and SSP team activities



#### Aerospace SSP Models





## Aerospace SSP Models -- Mass

(1.2 GW Delivered to Ground)

Cha-vada	Mass (MT)					
Subsystem	<b>Sun Tower</b>	Multi-Strand	POP	Halo	Laser	
Power Transmission	2,840	2,840	2,840	3,068	2.6 / 1,270	
Power Collection	4,964	4,964	4,972	3,648	10.0 / 4,784	
Attitude Control	170	170	170	58	2.4 / 1,128	
Backbone, Tether	7,994	2,385	1,253	43	1.1 / 545	
PMAD	6,131	6,117	6,108	5	0.5 / 223	
TT&C and C&DH	37	37	37	3	0.5 / 255	
Propulsion	3,216	3,217	3,297	2,988	5.2 / 2,499	
Thermal	944	944	944	698	0.8 / 390	
Robotics	200	200	200	200	0	
Dry Mass	26,497	20,874	19,820	10,712	23.1 / 11,094	
On-Orbit Propellant	1,044	1,045	1,434	1,275	0.4 / 180	
Transfer Propellant	1,986	1,984	2,059	3,172	6.2 / 2,984	
<b>Total Mass</b>	29,527	23,903	23,313	15,159	29.7 / 14,257	



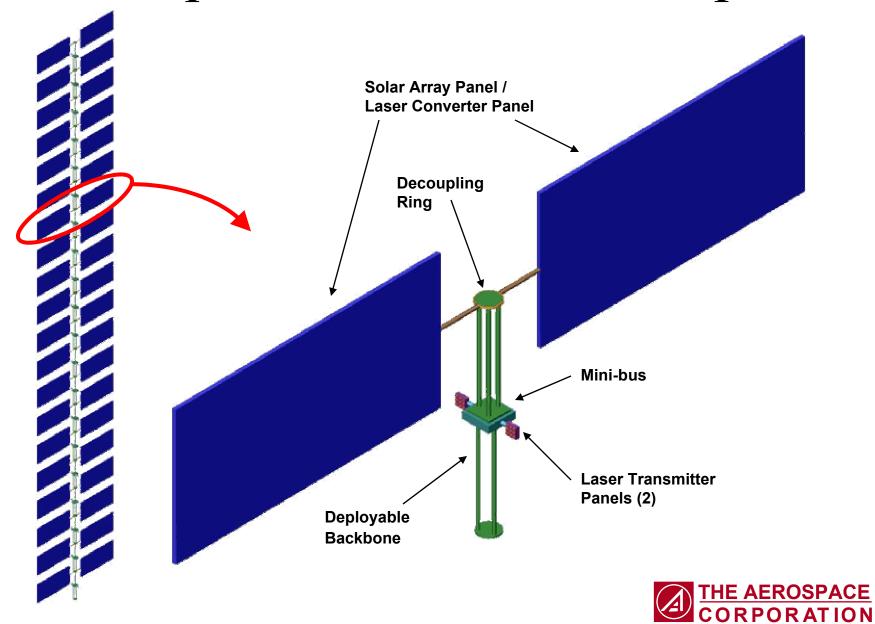
## Aerospace SSP Models -- Cost

(1.2 GW Delivered to Ground)

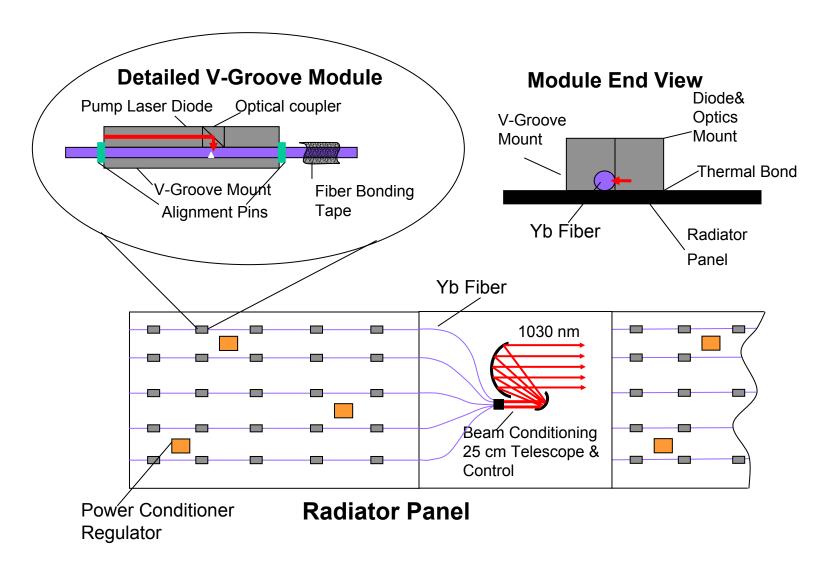
Cycle great area	Cost (\$FY00M)				
Subsystem	Sun Tower	<b>Multi-Strand</b>	POP	Halo	
Power Transmission	1,140	1,140	1,140	1,140	
Power Collection	8,447	8,447	8,447	5,752	
Attitude Control	505	505	505	53	
PMAD	532	320	252	2	
TT&C and C&DH	172	172	172	12	
Propulsion	7,880	7,881	7,885	5,421	
Propellant	1,000	1,000	1,153	1,468	
Structure	1,821	1,820	2,524	3,818	
Thermal	472	472	472	349	
Robotics	200	200	200	200	
System AIT&E	3,325	3,294	3,408	2,732	
Ground Support Eq.	5,320	5,270	5,453	4,371	
Software	516	516	516	516	
Program Management	5,464	5,412	5,599	4,491	
Spacecraft Cost	36,794	36,450	37,695	30,324	
Rectenna and Ground	5,156	5,156	5,156	5,156	
Launch	12,434	10,067	9,817	6,384	
<b>Total System Cost</b>	54,384	51,573	52,669	41,864	
\$/kW-hr for 30% IRR	\$1.08	\$1.02	\$1.04	\$0.83	



## Aerospace Laser SSP Concept

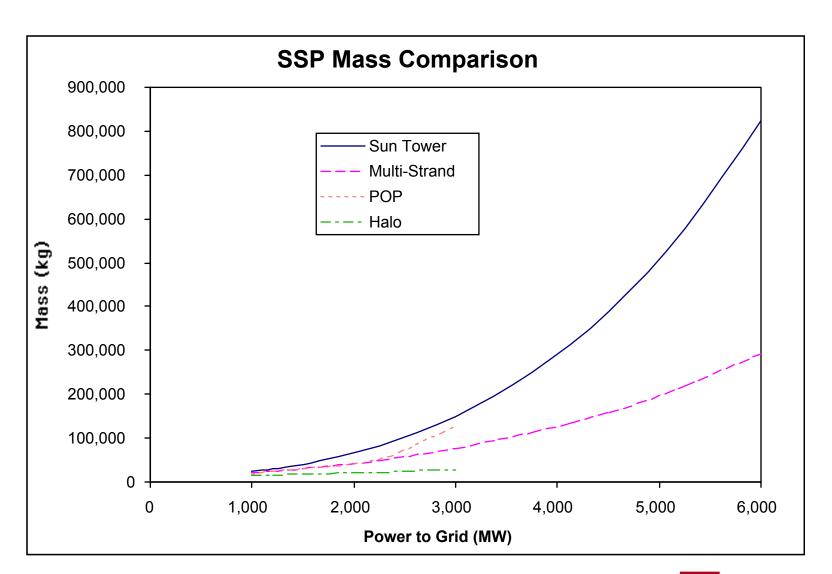


#### Optical Concept for Laser SSP



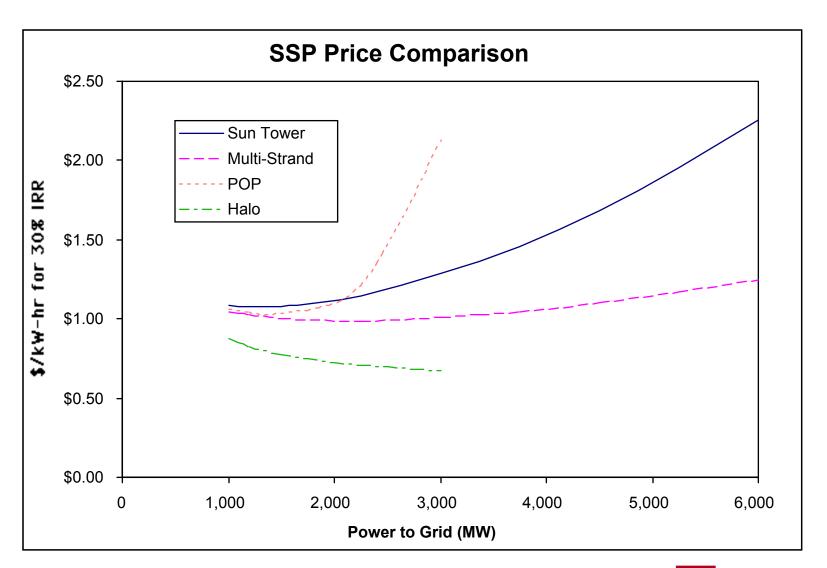


## Trade Space Exploration Example



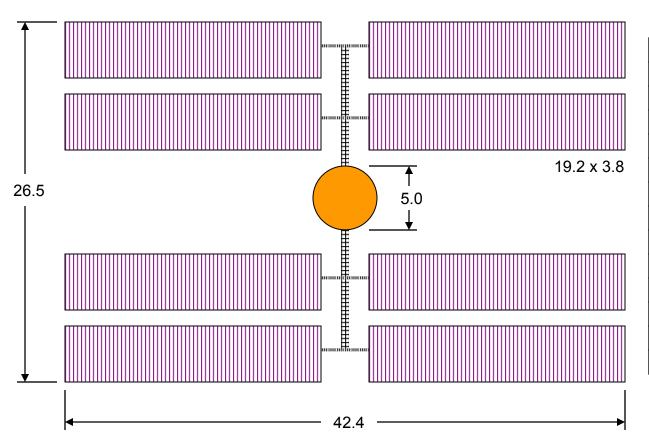


#### Trade Space Exploration Example





#### Microwave MSC-1 Concept Model

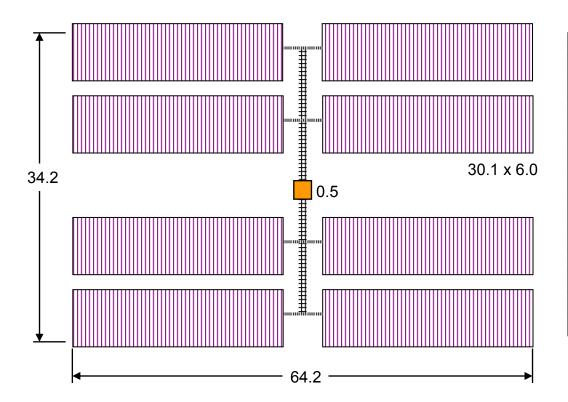


Γ <sub>2</sub>			
Subsystem	Mass (kg)		
Power Transmission	205		
Power Collection	4,413		
Attitude Control	157		
Backbone, Tether	408		
PMAD	3,560		
TT&C and C&DH	68		
Propulsion	1,284		
Thermal	7		
Robotics	2,068		
Dry Mass	12,167		
On-Orbit Propellant	506		
Transfer Propellant	3,411		
Launch Mass	16,084		

- MSC-1 model is dependent on Mission Delta-V
  - Can be resized to accommodate any science/commercial mission



## Laser MSC-1 Concept Model

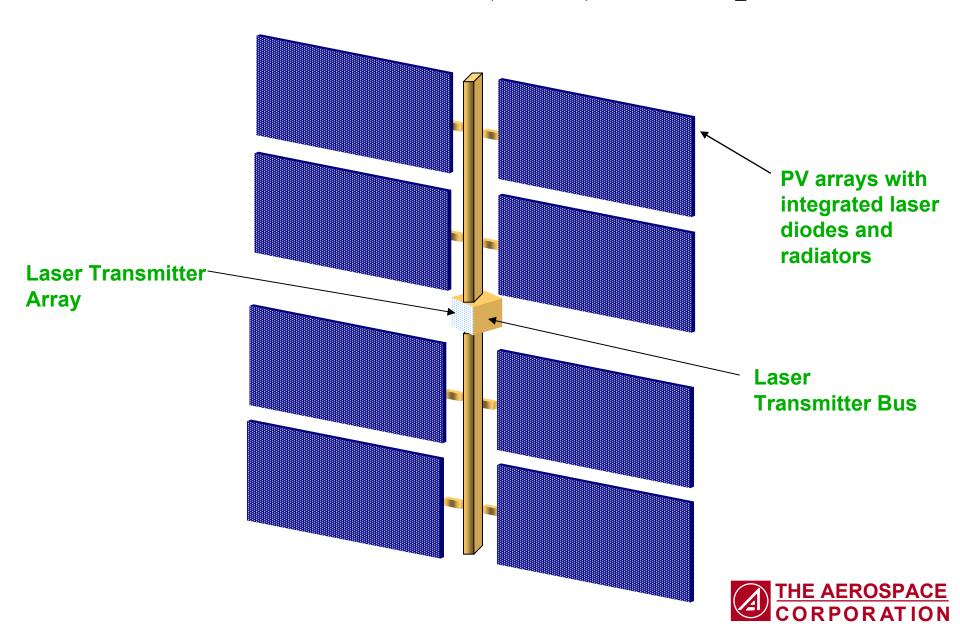


Subsystem	Mass (kg)		
Power Transmission	505		
Power Collection	6,700		
Attitude Control	157		
Backbone, Tether	1,023		
PMAD	36		
TT&C and C&DH	68		
Propulsion	1,225		
Thermal	424		
Structure	1,023		
Dry Mass	11,150		
On-Orbit Propellant	464		
Transfer Propellant	3,126		
Launch Mass	14,740		

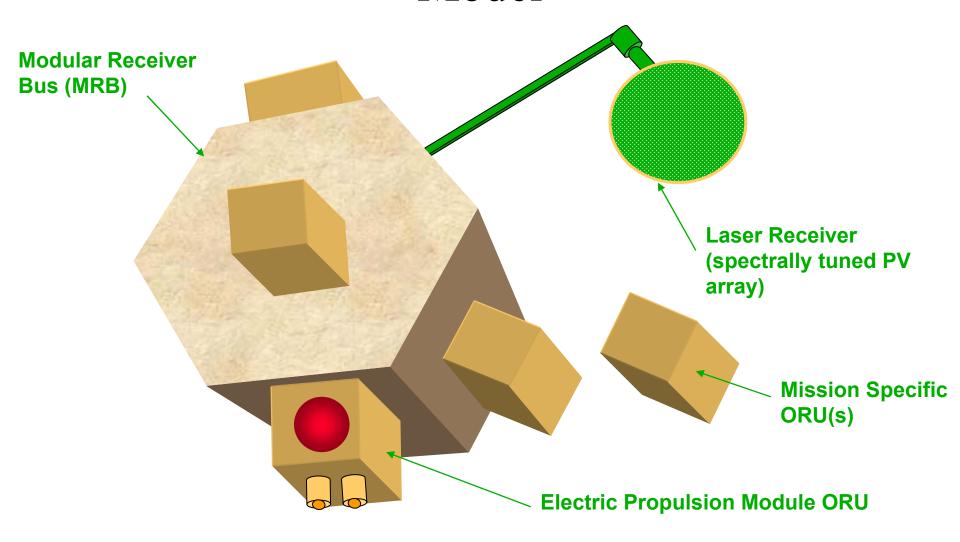
- MSC-1 model is dependent on Mission Delta-V
  - Can be resized to accommodate any science/commercial mission



#### Laser Transmitter Bus (LTB) Concept Model



#### Laser Modular Receiver Bus (MRB) Concept Model





#### Reference Mission Concept of Operations

- 1) Deploy and Checkout Laser SSP Transmitter Bus (LTB) and Modular Receiver Bus (MRB) DARPA NextSat Bus is an option for MRB
- 2) Demonstration of close proximity WPT from LTB to MRB
  - Provide calibration and characterization of Laser WPT
  - Use Electric propulsion thrusters or other high power device as the load
- 3) Modular Receiver Bus (MRB) uses electric propulsion to perform separation maneuver(s), vary separation range between 200 and 2000 miles demonstrates high power EP orbit transfer
- 4) Beam high power (100 kW class) beam to small diameter (10-12 meter) stationary site demonstrates power generation for terrestrial SSP
- 5) Beam moderate power (5-20 kW) to an electrically powered rover mounted with Laser receiver canopy
  - -Demonstrate planetary rover (or equipment) powered by orbital power station
- 6) Use Astro Bus to add Optional Orbital Replaceable Unit (ORU) to MRB -Add Mission Kit(s) 1-6 (described on next chart)

CONOPs designed to demonstrate SSP for terrestrial utility, space science/exploration and space industry applications

#### Potential ORU Mission Kits on MRB

- Baseline ORU Kit) Electric Propulsion Module containing EP thrusters, propellants, and power processing (launched with MRB and occupies one of multiple ORU bays on MRB)
- Kit 1) High Power Communications (Commercial, NASA Terrestrial, or NASA Planetary)
- Kit 2) Space Based Radar Mission Kit (Military and/or NASA Applications)
  - could potentially be a modular, autonomously assembled on orbit, phased array radar
- Kit 3) Manufacturing or High Power Experiment Module allows long duration zero G experiments or manufacture in proximity to ISS
  - -allows periodic replenishment or recovery of reactants, products, and other consumables either by Astro or by ISS crew/assets
- Kit 4) Space Based Laser or SSP Relay Mirror to demonstrate relay of laser energy over long terrestrial distances
- Kit 5) Microwave Transmitter Module: demonstrate transmission of 25 to 50 kW of microwave energy to Earth measure beam pattern to characterize microwave WPT
- Kit 6) Microwave Receiver Module: demonstrate high power near proximity (several miles) space-to-space microwave WPT transmission from Microwave SSP Demonstrator

#### Aerospace JIETSSP Tasks

- Define innovative concepts, applications, and orbits
  - Work with NASA to define set of SSP concepts to be modeled
- Develop/enhance SPS models
  - Enhance SSP concept models
  - Develop new models, based on chosen concepts and applications
  - Conduct subsystem-level trades
- Develop near-term flight demonstrator SPS models
  - Develop specific design details for the SSP demonstrator systems
  - Develop concept of operations for each demonstrator system
  - Conduct subsystem-level trades
- Conduct system-level trades and analysis SSP concepts and architectures
  - Compare various SSP concepts for similar applications
- Develop investment roadmaps based on lessons learned
  - Integrate the requirements of the most promising concepts into a set of technology goals
- Explore and develop synergies with other government efforts